# EXPERIMENTAL AND NUMERICAL INVESTIGATION OF COLD-FORMED LEAN DUPLEX STAINLESS STEEL FLEXURAL MEMBERS

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**Abstract:** Experimental and numerical investigation of cold-formed lean duplex stainless steel flexural members is presented in this paper. The test specimens were cold-rolled from flat plates of lean duplex stainless steel with the nominal 0.2% proof stress of 450 MPa. Specimens of square and rectangular hollow sections subjected to both major and minor axes bending were tested. A finite element model has been created and verified against the test results using the material properties obtained from coupon tests. It is shown that the model can accurately predict the behaviour of lean duplex stainless steel flexural members. An extensive parametric study was carried out using the verified finite element model. The test and numerical results as well as the available data on lean duplex beams are compared with design strengths predicted by various existing design rules, such as the American Specification, Australian/New Zealand Standard, European Code and direct strength method for cold-formed stainless steel. Reliability analysis was performed to evaluate the reliability of the design rules. It is shown that these current design rules provide conservative predictions to the design strengths of lean duplex stainless steel flexural members. In this study, modified design rules on the American Specification, Australian/New Zealand Standard, European Code and direct strength method are proposed, which are shown to improve the accuracy of these design rules in a reliable manner.

**Keywords:** Beam; Cold-formed steel; Design strength; Finite element modeling; Flexural members; Four-point bending tests; Hollow sections; Lean duplex; Stainless steel.

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#### 1. Introduction

Cold-formed stainless steel is gaining increasing applications as a construction material serving both architectural and structural needs. It provides aesthetic and modern shining appearance, superior corrosion resistance, longer service life with easy maintenance, and convenience in construction. Therefore, extensive research has been carried out on the structural performance of stainless steel structures. Design specifications for stainless steel structures were developed for various types of stainless steel, including ferritic, austenitic and duplex stainless steel. Nevertheless, the high cost of stainless steel material constrains its wider application. In recent years, a relatively new type of stainless steel, called lean duplex stainless steel of grade EN 1.4162 (LDX 2101), with structural and economical advantages was developed. It is becoming an attractive choice as a construction material due to its low cost compared to duplex stainless steel, and the strength of the material is comparable with duplex stainless steel. However, the lean duplex stainless steel is currently not covered in any design specification, and the investigation on such new material is also limited.

Theofanous and Gardner [1] carried out three-point bending tests on 8 specimens and finite element analysis on 36 specimens of lean duplex stainless steel rectangular hollow section (RHS) and square hollow section (SHS). It was found that the European Code is overly conservative, while the Australian/New Zealand Standard and the American Specification provided more accurate prediction to the strengths of flexural members. The modified classification limits that proposed by Gardner and Theofanous [2] and the continuous strength method (CSM) provided better prediction to the flexural members. Huang and Young [3] investigated the material properties of lean duplex stainless steel by conducting coupon tests, stub column tests and measurement of residual stresses. Column tests were conducted on cold-formed lean duplex stainless steel members by Huang and Young [4]. It was found that the current design specifications are generally conservative for columns, and a new design approach of using stub column property & full cross-sectional area in calculation compression capacity has been recommended. Furthermore, finite element analysis on lean duplex stainless steel columns was also performed by Huang and Young [5]. A total number of 259 column strengths were compared with design values predicted by various design rules. It is shown that the existing design rules are generally conservative. Modifications are proposed for the AS/NZS Standard, EC3 Code and direct strength method in order to obtain a more accurate prediction for the cold-formed lean duplex stainless steel columns. Saliba and Gardner [6] performed experimental and numerical investigation on the structural behaviour of lean duplex stainless steel welded I-sections. The investigation included coupon tests, stub column tests and bending tests as well as parametric study on welded I-sections using finite element analysis. The experimental and numerical data were compared with design predictions by European Code for stainless steel and continuous strength method (CSM). It is shown that the current Class limits in the European Code can be relaxed. In addition, the continuous strength method is shown to provide better prediction than the current European Code prediction.

The objective of this study is mainly to investigate the structural performance of cold-formed lean duplex stainless steel flexural members. A series of bending tests and a wide range of parametric study on lean duplex stainless steel flexural members were carried out. The 180 numerical and experimental data obtained from this study and previous research [1] were compared with design predictions by the American Specification (ASCE) [7], Australian/New Zealand Standard (AS/NZS) [8], European Code (EC3) [9], the design rule proposed by Gardner and Theofanous [2] and the direct strength method (DSM) described in the North American Specification (AISI) [10]. Reliability analysis was conducted for each of the current design rules, and design recommendations are proposed in this study. It should be noted that the lean duplex stainless steel is not covered in the ASCE, AS/NZS nor EC3.

# 2. Experimental Investigation

#### 2.1 Test specimens

Cold-formed lean duplex stainless steel flexural members were tested subjected to pure bending. The nominal 0.2% proof stress of the lean duplex stainless steel is 450 MPa. There are six different sections including two square hollow sections (SHS) and four rectangular hollow sections (RHS). The test specimens used in this study are the same batch of specimens as those investigated by Huang and Young [3]. The material properties as shown in Table 1 are also reported by Huang and Young [3]. The specimens of RHS were tested under bending about both the major and minor axes. The specimens were labelled such that the depth of the web (D), width of the flange (B), thickness (t) of the cross-section as well as the specimen length (L) can be recognized. The arrangement of the cross-sectional dimensions also refers to the bending axis. For example, the label  $30 \times 50 \times 2.5$ L900 defines the following specimen. The numbers before the letter "L" is refer to the cross-sectional dimension. The dimensions of the web (D), flange (B) and thickness (t) of the cross-section are equal to 30, 50 and 2.5 mm, respectively. The numbers after the letter "L" indicates the specimen length of 900 mm.

The dimension of the web (D) is smaller than the flange (B), thus the beam is subjected to minor axis bending. On the other hand, the specimen  $50\times30\times2.5$ L900 is of the same cross-sectional dimension and length, but subjected to major axis bending.

#### 2.2 Test setup and procedure

The four-point bending tests were conducted to obtain the moment capacity of each test specimen. The relationship between the bending moment and curvature of the specimens can also be obtained. A total of ten four-point bending tests was conducted. The test setup is shown in Fig. 1. The pin-ended boundary conditions were simulated by a half-rounded support and a roller support located at 70 mm from the two ends of the specimen. Major and minor axes bending tests were carried out on the specimens of RHS. The specimens of SHS were placed so that the surface with the weld is located at the web of the sections. The moment span between the two loading points, and the shear span between the end supports and the loading points were carefully designed, so that the section moment capacity could be obtained without the occurrence of shear failure. Vertical loading was applied through a lockable ball bearing connecting to a spreader beam. The function of the lockable ball bearing is to eliminate any possible gaps between the spreader beam and the two loading points. The bearing was locked by four bolts and restrained from rotation prior to testing, as shown in Fig. 1.

Web stiffening plates were clamped at the two loading points and the ends of each specimen. In addition, wooden blocks were inserted at these locations to prevent any possible local bearing failure during testing. Three displacement transducers (LVDTs) were placed along the centerline of the tension face of each specimen at the two loading points and at the mid-span of the specimen. The vertical deflections of the specimen at these three locations (two loading points and mid-span) were recorded, and the curvature of the specimen was calculated from the recorded deflections. Concentrated compressive force was applied by a hydraulic testing machine using displacement control with a constant loading rate of 1.0 mm/min for all test specimens. The static load was recorded by pausing the applied staining for two minutes at the ultimate load. A data acquisition system was used to record the applied load and the readings of the LVDTs at regular intervals during the tests.

#### 2.3 Test results

The experimental ultimate moments ( $M_{Exp}$ ) and the corresponding curvatures ( $k_{Exp,u}$ ) of the test specimens are summarized in Table 3. The static moment-curvature curve for each specimen is plotted in Fig. 2(a). The static moment (M) and the curvature (k) of each specimen are normalized with the plastic moment ( $M_{pl}$ ) and the curvature corresponding to plastic moment ( $k_{pl}$ ), respectively, as shown in Fig. 2(b). The moments were obtained using half of the static applied load from the actuator multiplied by the shear span of the specimens. Out-of-plane bending was not observed in the tests. In addition to flexural behaviour (F), local buckling (L) is also observed at ultimate load of specimens  $50 \times 50 \times 1.5$ L900,  $50 \times 70 \times 2.5$ L1100,  $50 \times 100 \times 2.5$ L1500 and  $50 \times 150 \times 2.5$ L1500. The failure modes observed at ultimate load of the specimens  $50 \times 30 \times 2.5$ L900 and  $50 \times 50 \times 1.5$ L900 involved flexural behaviour (F) and combination of flexural behaviour and local buckling (L+F) are shown in Figs 3 and 4, respectively. The curvatures ( $k_{Exp,u}$ ) of test specimens were calculated from displacements measured from the three LVDTs. A constant curvature between the transducer locations was assumed, and the curvature was calculated using the radius (r) of the curved beam specimen between the LVDTs located at the two loading points, such that  $k_{Exp,u} = 1/r$ .

The experimental ultimate moments ( $M_{Exp}$ ) are compared with the theoretical elastic ( $M_{el}$ ) and plastic ( $M_{pl}$ ) bending moments, as shown in Table 3. The elastic and plastic bending moments were calculated using the measured 0.2% proof stress ( $\sigma_{0.2}$ ) obtained from the flat coupon tests, as shown in Table 1, multiplied by the elastic and plastic section moduli of the full sections, respectively. Generally, conservative predictions to moment capacity of the test specimens were found, especially for those subjected to major axis bending. The mean value of  $M_{Exp}/M_{el}$  and  $M_{Exp}/M_{pl}$  ratios is equal to 1.33 and 1.10 with the corresponding coefficients of variation (COV) of 0.206 and 0.178, respectively. The moment capacity of specimen  $50 \times 150 \times 2.5 \text{L}1500$  was over predicted by elastic and plastic bending moments with the  $M_{Exp}/M_{el}$  and  $M_{Exp}/M_{pl}$  ratios of 0.76 and 0.68, respectively. This is due to the early occurrence of local buckling in the compression flange before the specimen reached yielding.

#### 3. Finite element model

Finite element model was developed using the program ABAQUS version 6.11 [11] to simulate the cold-formed lean duplex stainless steel flexural members. The initial local geometric imperfections and materials properties obtained from tensile coupon tests of flat

portions and corners measured by Huang and Young [3] were incorporated in the finite element model. A four-noded doubly curved shell element with reduced integration (S4R) with a mesh size of  $10 \text{ mm} \times 10 \text{ mm}$  (length by width) in the flat portions of the cross-sections and a finer mesh at the corners were used.

In the experiments, the concentrated compressive load was applied vertically through the lockable bearing and spreader beam, and then the load transferred to the roller and half-rounded bar onto the load transferring plates to the specimen, as shown in Fig. 1. Therefore, the two loading points were modelled by two reference points located at the middle of the contact surfaces between the load transferring plates and the specimen. The reference points were coupled to the contact surfaces between the transferring plates and the specimen, and restrained against all degrees of freedom except for displacement in the vertical and longitudinal directions along the flexural member as well as the rotation about the bending axis. Similarly, the two supports were modelled by coupling the contact surfaces with two reference points located at the bottom flange of the specimen in the middle of the corresponding contact surfaces. The pin support (half-rounded) was modeled by restraining against all degrees of freedom except for the rotation about the bending axis, while the roller support was modeled by allowing an extra degree of freedom for longitudinal displacement along the specimen. The loading was applied by displacement control method, which is identical to the tests of flexural members, by specifying an axial displacement at the two reference loading points. In the finite element model, the loading was applied by a static RIKS step available in the ABAQUS library. The nonlinear geometric parameter (\*NLGEOM) was included to deal with the large displacement analysis.

Huang and Young [3] conducted the tensile coupon tests to obtain the material properties of the cold-formed lean duplex stainless steel specimens. These specimens are of the same batch as the beam specimens in this study. The coupons were extracted from the flat portions and corners of each section, and the measured stress-strain curves were used in the finite element model. The material properties including the 0.2% proof strength (yield strength) ( $\sigma_{0.2}$ ), ultimate strength ( $\sigma_u$ ), strain at fracture ( $\varepsilon_f$ ), initial Young's modulus ( $E_o$ ), and Ramberg-Osgood parameter (n) of flat and corner coupon tests for each section are summarized in Table 1. A multi-linear stress-strain curve containing the elastic part up to the proportional limit stress with the measured Young's modulus, and the plastic part with the true stress and logarithmic true plastic strain curve, which is converted from a static stress-strain curve, was used. The true plastic stress-strain curves converted from flat coupon test results were used as the material properties in modelling the flat portions of the

specimens, while those converted from the corner coupon test results were used in modelling the corner regions of the specimens.

The local geometric imperfections for each section that measured by Huang and Young [3] were included in the finite element model. The local buckling mode, which was obtained by carrying out Eigenvalue analysis with a large D/t ratio and using a BUCKLE procedure, was superposed on the flexural member model. The first buckling mode predicted by the ABAQUS Eigenvalue analysis are normalized to 1.0, thus the buckling mode was factored by the measured magnitudes of the initial local geometric imperfections for each member.

In addition, the membrane residual stresses of section  $150 \times 50 \times 2.5$ , which was measured by Huang and Young [3], was included in the model of specimens  $150 \times 50 \times 2.5 L1500$  and  $50 \times 150 \times 2.5 L1500$  using the ABAQUS (\*INITIAL CONDITIONS, TYPE = STRESS) parameter to assess the influence of the residual stresses on the beam capacities of cold-formed lean duplex stainless steel flexural members. The section was partitioned into strips of the same width as those measured by Huang and Young [3]. It should be noted that the bending residual stresses effect has been taken into account by the material properties obtained from flat and corner coupon tests.

The ultimate moments predicted by finite element analysis ( $M_{FEA}$ ) including residual stresses are compared with the test results ( $M_{Exp}$ ). The numerical results obtained by the model excluding residual stresses are also compared with the test results. The moment ratio  $M_{Exp}/M_{FEA}$  of specimen  $150\times50\times2.5L1500$  equals to 0.96 for the finite element model includes residual stresses, while the  $M_{Exp}/M_{FEA}$  ratio equals to 0.94 for the model without considering residual stresses. It is shown that the finite element model including residual stresses is 2% more accurate than that without considering residual stresses. For the specimen  $50\times150\times2.5L1500$ , the  $M_{Exp}/M_{FEA}$  ratios are both equal to 1.00 for the finite element models with and without residual stresses. The finite element model with residual stresses is 0.3% more accurate than that without residual stresses. The moment-curvature curves of the test and finite element analysis for specimen  $50\times150\times2.5L1500$  are shown in Fig. 5. It is shown that the effect of residual stresses on the beam are quite small, thus the residual stresses are not included in the parametric study.

#### 4. Verification of Finite Element Model

The moment capacities ( $M_{FEA}$ ) predicted by the finite element analysis are compared with the test results ( $M_{Exp}$ ), as shown in Table 3. The mean value of  $M_{Exp}/M_{FEA}$  ratio is 1.00 with the coefficient of variation (COV) of 0.040. A maximum difference of 7% is found between the experimental and numerical moment capacities for specimen  $50\times50\times1.5$ L900. The failure modes predicted by the finite element analysis are identical to those obtained from the tests at ultimate load. The failure modes of flexural buckling (F) and interaction of local and flexural buckling (L+F) obtained from the FEA compare well with the experimental failure modes for specimens  $50\times30\times2.5$ L900 and  $50\times50\times1.5$ L900 as shown in Figs 3 and 4, respectively. The comparison of curvatures of the test specimens ( $k_{Exp,u}$ ) with those of the finite element results are summarized in Table 3. The mean value of  $k_{Exp,u}$ / $k_{FEA,u}$  equals to 1.04 with the COV of 0.199. Fig. 5 shows the comparison of the experimental and numerical results for specimen  $50\times150\times2.5$ L1500. It is observed that good agreement has been achieved in terms of the ultimate flexural strength and corresponding curvature.

# 5. Parametric Study

An extensive parametric study was performed using the verified finite element model with a total of 126 flexural members of lean duplex stainless steel. The material properties adopted in the parametric study were based on the stress-strain curves obtained from the flat and corner coupon tests of section  $70\times50\times2.5$  reported in Huang and Young [3]. The averaged measured local geometric imperfection for the tested specimens reported in Huang and Young [3, 4] was t/11, where t is the thickness of each section. Thus, a slightly conservative rounded number of t/10 was used as the local imperfection in the parametric study. The residual stresses of the flexural members are not included in the finite element model, considering its negligible effect on the moment capacity.

The 126 specimens in the parametric study were SHS and RHS, with 6 different overall profiles (overall depth  $\times$  overall width) of SHS ranged from 50×50 to 300×300, and 7 different overall profiles of RHS ranged from 50×30 to 350×100. The thicknesses of each profile varied to cover a wide range of slenderness ratio from stocky to slender sections. The aspect ratio (D/B) for the specimens was ranged from 0.25 to 4. The moment span between the two loading points was 500 mm for all specimens, and the shear span between the loading points to the supports were carefully designed so that the section flexural capacity can be

reached without shear failure. The RHS specimens were subjected to both major and minor axes bending. The specimens in the parametric study used the same labelling system as that of the test specimens, as shown in Table 4. The ultimate moment capacities and the corresponding curvatures predicted by the finite element analysis are summarized in Table 4.

# 6. Reliability Analysis

The suitability of the current design rules, including ASCE [7], AS/NZS [8], EC3 [9], modified EC3 by Gardner and Theofanous [2], direct strength method (DSM) in the AISI [10] and continuous strength method (CSM) [6] for the cold-formed lean duplex stainless steel flexural members is evaluated using reliability analysis, which is detailed in the Commentary of the ASCE Specifications [7]. A target reliability index ( $\beta_0$ ) of 2.5 for stainless steel structural members is used as a lower limit. The design rules are considered to be reliable if the reliability index is greater than or equal to 2.5. The resistance factors  $(\phi_0)$  of 0.90 for members with stiffened compression flanges subjected to bending is recommended by ASCE [7], AS/NZS [8], and AISI Standard [10] for direct strength method (DSM), while the resistance factors of 0.91 is used by the EC3 [9], the modified EC3 by Gardner and Theofanous [2] as well as the continuous strength method (CSM) [6]. The load combinations of 1.2DL+1.6LL, 1.25DL+1.5LL and 1.35DL+1.5LL were used for design rules in ASCE, AS/NZS and EC3 in the reliability analysis, respectively, where DL is the dead load and LL is the live load. The load combination of 1.35DL+1.5LL was used for reliability analysis of modified EC3 by Gardner and Theofanous [2] and continuous strength method (CSM), while the load combination of 1.2DL+1.6LL was used for the direct strength method (DSM). The Eq. 6.2-2 in the ASCE Specification [7] was used in calculating the reliability index. The statistical parameters  $M_m = 1.10$ ,  $F_m = 1.00$ ,  $V_m = 0.10$  and  $V_F = 0.05$ , which are the mean values and coefficients of variation for material properties and fabrication factors for flexural members in Clause 3.3.1.1 of the commentary of the ASCE Specification were adopted. The mean value  $(P_m)$  and coefficient of variation  $(V_P)$  of tested-to-predicted load ratio or numerical results to design predictions ratio are shown in Table 5. In calculating the reliability index, Eq. F1.1-3 in the North American cold-formed steel Specification AISI S100 [10] was used to calculate the correction factor, in order to take into account for the influence by the number of data. For the purpose of direct comparison, a constant resistant factor  $(\phi_1)$  of 0.90 and a load combination of 1.2DL+1.6LL were used to calculate the

reliability index ( $\beta_I$ ) for the design rules, and the values of the reliability index are also shown in Table 5.

# 7. Current Design Rules and Comparison of Moment Capacities

#### 7.1 General

The experimental and numerical moment capacities ( $M_u$ ) are compared with the *unfactored* design flexural strengths (nominal strength) predicted by ASCE [7], AS/NZS [8], EC3 [9], modified EC3 by Gardner and Theofanous [2], DSM [10] and CSM [6]. The comparison of the experimental and numerical moment capacities with the design moment capacities is shown in Table 5 and 6. The flat coupon test result of section  $70\times50\times2.5$ , reported in Huang and Young [3], is used in calculating the design flexural strengths for specimens in the parametric study. The design rules in ASCE [7], AS/NZS [8], EC3 [9] and modified EC3 by Gardner and Theofanous [2] used the effective width method for the sections when local buckling occurs. Therefore, the calculation procedure using these design rules involved iterative process, as the location of the neutral axis shifts with the effective width when the sections subjected to bending. However, such tedious iterative process is not required in the DSM [10] and CSM [6], as the flexural strength is calculated by the full section instead of effective section. It should be noted that the ASCE, AS/NZS and EC3 do not cover the material of lean duplex stainless steel.

#### 7.2 American Specification and Australian/New Zealand Standard

The ASCE [7] and AS/NZS [8] use the same design rules to calculate the moment capacity and the effective width of the section. According to Clause 3.3.1.1 in the ASCE Specification and Clause 3.3 of the AS/NZS Standard, the two design specifications allow the calculation base on initiation of yielding and inelastic reserve capacity. Therefore, both approaches are assessed in this study.

For the approach by initiation of yielding, the moment capacities ( $M_{yielding}$ ) were calculated by the effective section modulus ( $S_e$ ) multiplied by the yield strength ( $f_y$ ). The effective width was calculated in accordance with Clause 2.2 in ASCE and AS/NZS, where a yield strength at the extreme fibre of compressive flange and the stress distribution vary linearly in the

section were assumed. It is shown in Table 5 that the initiation of yielding approach in ASCE [7] and AS/NZS [8] provide quite conservative and scattered predictions for moment capacities of the specimens, especially to those stocky sections with low slenderness ratio  $d/(t\varepsilon)$ , where d is the flat portion of the web and  $\varepsilon$  is the material factor, as shown in Table 5 and Fig. 6. The mean value of the  $M_u/M_{yielding}$  ratio is 1.50 and the coefficient of variation (COV) equals to 0.264. This approach is considered to be reliable for ASCE [7] and AS/NZS [8] with the reliability indices ( $\beta_0$ ) of 2.85 and 2.71, respectively.

For the approach by inelastic reserve capacity, the moment capacities ( $M_{inelastic}$ ) were calculated by the equivalent force multiplying the lever arm within the section, considering equilibrium of stresses in the effective section and assuming an ideally elastic-plastic stress distribution in the section. The compression strain factor ( $C_y$ ) was calculated to determine the stress distribution in the section. However, the ASCE [7] and AS/NZS [8] did not state clearly the calculation of effective widths that involve elastic-plastic stress distribution in the section. According to Yu and Laboube [12] for elastic-plastic stress distribution in the section, the effective widths were calculated using the equations for elements with stress gradient in Clause 2.2.2 of the ASCE [7] and AS/NZS [8], as shown in Fig. 7(a). It is required in both the ASCE [7] and AS/NZS [8] that the ratio of the depth of compressed portion of the web to its thickness does not exceed the slenderness ratio. In this study, there are 47 out of 180 specimens that exceeded this limitation. Thus, 133 specimens within the limit were compared using the inelastic reserve capacity design approach, as shown in Table 5.

It is shown that the mean value of  $M_u/M_{inelastic}$  ratio is 1.34 and the COV equals to 0.200, which is less conservative and less scattered than the approach by initiation of yielding. The reliability indices ( $\beta_0$ ) for the ASCE [7] and AS/NZS [8] are 2.92 and 2.76, respectively, which are larger than the target value of 2.50. The comparison of experimental and numerical results with design strengths by ASCE [7] and AS/NZS [8] is shown in Fig. 6, where the flat width b is obtained by the flat portion of the flange. It should be noted that both approaches provide quite conservative predictions to specimens with the compression strain factor ( $C_y$ ) equal to 3, which is the elastic-plastic stress distribution occurred in the section. The comparison of the test and numerical results with the design predictions to the 54 specimens with compression strain factor equal to 3 are shown in Table 6. It is shown that the approach by inelastic reserve capacity is very conservative for these specimens with the mean value of  $M_u/M_{inelastic}$  equal to 1.54 and COV of 0.179 for both ASCE [7] and AS/NZS [8]. Therefore, design rules for stainless steel flexural members need to be modified.

Furthermore, the limitation of the approach by inelastic reserve capacity, that the ratio of the depth of the compressed portion in the web  $(d_w)$  to its thickness does not exceed the slenderness ratio  $(\lambda_l)$ , was assessed by comparing the design flexural strengths  $(M^*_{inelastic})$  of the 47 specimens which exceeded the limitation  $(d_w/t > \lambda_l)$  with the test and numerical results, as shown in Table 5, where  $\lambda_l$  is defined in Clause 3.3 of ASCE [7] and AS/NZS [8]. The result shows that the approach provides good prediction on these specimens even they are beyond the limitation. The mean value of  $M_u/M^*_{inelastic}$  is 1.17 with the COV of 0.153, and the reliability indices  $(\beta_0)$  of 2.72 and 2.55 for ASCE [7] and AS/NZS [8], respectively.

# 7.3 European Code

According to Clause 5.1 of the EC3 Part 1.4 [9], the provisions given in Section 5 and 6 of EC3 Part 1.1 [13] should be applied for stainless steel, except where modified or superseded by the special provisions given in EC3 Part 1.4. Therefore, the moment capacity  $(M_{EC3})$  was calculated by the equations 6.13 to 6.15 of Clause 6.2.5 in EC3 Part 1.1 [13]. Classification was required according to Table 5.2 in EC3 Part 1.4, where the flat portions of the web and compressive flange were classified since these elements subjected to bending and compression. For specimens classified as Class 4 section, the effective widths were calculated by Table 4.1 of EC3 Part 1.5 [14] together with the reduction factor calculated by Clause 5.2.3 in EC3 Part 1.4. The classification is governed by the larger Class of element in the section. In this study, 80 specimens had a larger class in the flange compared to the class in the web, while 24 specimens had a larger class in the web instead of the flange. The remaining 76 sections had the same class in either the flange or web. It should be noted that there are not many investigation covered lean duplex stainless steel flexural members for section having a larger class in the web. The 24 specimens from the parametric study are used to assess the classification of elements subject to bending in the EC3 Code. The comparison of the experimental and numerical data with the design strengths is shown in Fig. 8 and Table 5. It is shown that the EC3 predictions are quite conservative with the mean value of  $M_u/M_{EC3}$ equal to 1.25 and COV of 0.147. The reliability index ( $\beta_0$ ) is 2.83, which is greater than the target reliability index of 2.5.

Gardner and Theofanous [2] analysed the classification limits of the European Code [9] for stainless steel. A set of classification limits for compression elements was proposed, and the effective width equations [2] were modified as shown in Eqs. (1-2):

$$\rho = \frac{1}{\lambda_p} - \frac{0.188}{\lambda_p^2} \le 1 \qquad \text{for outstand elements}$$
 (1)

$$\rho = \frac{0.772}{\overline{\lambda}_p} - \frac{0.079}{\overline{\lambda}_p^2} \le 1 \qquad \text{for internal elements}$$
 (2)

where  $\rho$  is the reduction factor for local buckling and  $\overline{\lambda_p}$  is the element slenderness. In this study, the design moment capacities  $(M_{G\&T})$  predicted by Gardner and Theofanous [2] in modifying the European Code were compared with the test and numerical results obtained from this study and available data. It is shown that the modified EC3 by Gardner and Theofanous [2] provides a more accurate and less scattered prediction, with the mean value of  $M_u/M_{G\&T}$  equal to 1.19 with the COV of 0.133 as well as the reliability index  $(\beta_0)$  of 2.73.

In European Code, a section is classified as Class 1 when it reaches plastic moment capacity for full section and be able to maintain sufficient deformation capacity, which is calculated by Eq. (3),

$$R = \frac{k_{pl}^{\hat{}}}{k_{pl}} - 1 \tag{3}$$

where  $k_{pl}$  is the curvature corresponding to the plastic moment  $(M_{pl})$  on the ascending branch of moment-curvature curve, while  $k_{pl}^{\hat{}}$  is the curvature on the descending branch at the plastic moment after the ultimate moment. The locations of  $k_u$ ,  $k_{pl}^{\hat{}}$  and  $k_{pl}$  on the moment-curvature curve of specimen  $100\times50\times2.5L1500$  is shown in Fig. 8. For specimens  $30\times50\times2.5L900$ ,  $50\times30\times2.5L900$ ,  $50\times50\times2.5L900$  and  $50\times70\times2.5L1100$ , the curvature  $(k_{pl}^{\hat{}})$  was not recorded, because the deformations of the specimens were very large and tests had to stop before reaching  $k_{pl}^{\hat{}}$ . Therefore, the maximum recorded curvatures for these four specimens were used as the value of  $k_{pl}^{\hat{}}$  in Eq. (3) to calculate the rotation capacities (R). Since there is no required deformation capacity for Class 1 in the EC3 Part 1.4 [9], the requirement of R=3 for carbon steel are adopted in this study to assess the Class 1 limit [15, 16], and Gardner and Theofanous [2] also adopted the same requirement for stainless steel. The rotation capacity R of the specimens with moment capacity exceeds the plastic moment capacity are plotted against the slenderness of the flange  $(b/(t\epsilon))$  and web  $(d/(t\epsilon))$  in Figs 10 and 11, respectively. The specimens governed by the flange are those with a larger class of the flange than the web, as plotted in Fig. 10, while the specimens governed by the web are those with a larger class of

the web than the flange, as plotted in Fig. 11. It is shown in Figs 10 and 11 that the current Class 1 limits for carbon steel in EC3 Part 1.1 [13] is more appropriate than those for stainless steel in EC3 Part 1.4 [9], as the limit of moment capacity exceeds the plastic moment while maintaining rotation capacity *R* larger than 3, in designing for lean duplex stainless steel internal elements subjected to compression and bending. A section is classified as Class 2 and Class 3 when it reaches plastic and elastic moment capacities, respectively. Therefore, the moment capacities obtained from the test and numerical results are normalized with plastic moment capacity for Class 2 and elastic moment capacity for Class 3 that plotted against the flange and web slenderness, in order to assess the Class 2 and Class 3 limits in EC3 Code, as shown in Figs 12 to 15. It is observed that the current Class 2 and Class 3 limits for stainless steel are generally conservative for lean duplex stainless steel beams. Hence, the current Class 2 and Class 3 limits for stainless steel can be relaxed for lean duplex material.

# 7.4 Direct Strength Method

The direct strength method used in this study was based on the clause 1.2.2 of Appendix 1 in the North American Specification for the Design of Cold-Formed Steel Structural Members [10]. The nominal flexural strength  $(M_{DSM})$  shall be determined by the minimum of the nominal flexural strength for lateral-torsional buckling  $(M_{ne})$ , local buckling  $(M_{nl})$  and distortional buckling  $(M_{nd})$ . The lateral-torsional buckling and distortional buckling did not occur for SHS and RHS. According to the Clause 1.2.2.2 in the Appendix 1 of Commentary on the North American Cold-formed Steel Specification [17], for fully braced beams which are restrained against lateral-torsional buckling, the maximum  $M_{ne}$  value is taken as the yield moment  $(M_y)$ . According to Fig. C-1.2.2-1 in the Appendix 1 of Commentary on the North American Cold-formed Steel Specification [17], the flexural strength for local buckling  $(M_{nl})$  is calculated by Eqs. 1.2.2-5, 1.2.2-6 and 1.2.2-7 in the North American Cold-formed Steel Specification [10], for which the nominal flexural strength for lateral-torsional buckling  $(M_{ne})$  is replaced by the yield moment  $(M_y)$ . Therefore, the nominal flexural strength  $(M_{DSM})$  equation [10] is shown in Eq. (4):

$$M_{DSM} = \begin{cases} M_y & \text{for } \lambda_l \le 0.776 \\ \left(1 - 0.15 \left(\frac{M_{crl}}{M_y}\right)^{0.4} \left(\frac{M_{crl}}{M_y}\right)^{0.4} M_y & \text{for } \lambda_l > 0.776 \end{cases}$$

$$(4)$$

where the yield moment  $(M_y)$  is equal to  $S_f f_y$ , and  $S_f$  = gross section modulus,  $f_y$  = yield strength, and  $\lambda_l = (M_y/M_{crl})^{0.5}$ . The critical elastic local buckling moment  $(M_{crl})$  of the cross-section was obtained from a rational elastic finite strip buckling analysis [18] with a 5 mm half-wave length interval.

It is shown that the DSM also provided conservative and scattered predictions to the flexural members of SHS and RHS considered in this study, as shown in Table 5. The mean value of  $M_{u}/M_{DSM}$  ratio is 1.35 with the corresponding COV and the reliability index ( $\beta_0$ ) of 0.172 and 3.12, respectively. The comparison of test and numerical results with design strengths by DSM is also shown in Fig. 16. It is found that the DSM provides a very conservative prediction for the 81 specimens with  $\lambda_l$  smaller than 0.776, as shown in Table 6. The mean value of  $M_{u}/M_{DSM}$  ratio is 1.47 with COV of 0.138. On the other hand, the DSM provides good prediction to the 99 specimens with  $\lambda_l$  greater than or equal to 0.776, with the mean value of  $M_{u}/M_{DSM}$  ratio equals to 1.16 and COV of 0.084. Therefore, the DSM for specimens with  $\lambda_l$  small than 0.776 should be modified, in order to provide a more accurate prediction for cold-formed lean duplex stainless steel flexural members.

#### 7.5 Continuous strength method

The continuous strength method (CSM) is a deformation-based design method. Gardner and Theofanous [2] and Saliba and Gardner [6] have shown that the CSM is capable of providing accurate predictions for stainless steel flexural members. Similar to the DSM, the cross-section classification and effective width calculation are not required in CSM. The continuous strength method used in this study was based on the CSM equations for bending resistances presented in Saliba and Gardner [6]. In addition, the CSM allows for strain hardening in determining the cross-section resistance. It is stated in Saliba and Gardner [6] that the CSM does not apply to cross-sections where the slenderness  $(\overline{\lambda}_p)$  larger than 0.748, because there is no significant benefit to be derived from strain hardening beyond this limit [6]. Therefore, the flexural strengths of 96 specimens that meet the requirement of the CSM are compared with the design values  $(M_{CSM})$  by the continuous strength method, as shown in Fig. 5. It is found that the CSM provides the best prediction among the existing design rules. The mean value of  $M_u/M_{CSM}$  ratio is 1.13 with COV of 0.087. The reliability index  $(\beta_0)$  is equal to 2.75, which is greater than the target value of 2.5.

It is noted that the continuous strength method is capable of providing good prediction for the cold-formed lean duplex stainless steel flexural members, while the other existing design rules investigated in this study provide quite conservative predictions for the moment capacities of lean duplex stainless steel flexural members. Therefore, modifications to the ASCE and AS/NZS as well as EC3 and DSM are proposed in the following section.

# 8. Modified Design Rules & Comparison of Beam Strengths

#### 8.1 General

The ASCE Specification [7], AS/NZS Standard [8], EC3 Code [9] and DSM [10] do not cover the design of lean duplex stainless steel. It was found that these design rules provide quite conservative predictions to the moment capacity of lean duplex stainless steel. Therefore, modifications to these design rules are proposed for lean duplex stainless steel flexural members. A total number of 180 test and numerical results as well as the available data was used in the modifications of the design rules. The design strengths calculated by the modified inelastic reserve capacity approach in the ASCE Specification and AS/NZS Standard as well as the modified EC3 Code and DSM are represented by  $M^{\#}_{inelastic}$ ,  $M^{\#}_{EC3}$ , and  $M^{\#}_{DSM}$ , respectively.

#### 8.2 ASCE Specification and AS/NZS Standard

The approach by initiation of yielding is considered to be too conservative in designing flexural members of lean duplex stainless steel, due to the assumption of linear elastic stress distribution in the section for all specimens including those reached elastic-plastic stress distribution. In this study, modifications are made to the approach by inelastic reserve capacity, especially to those stocky sections with compression strain factor ( $C_v$ ) equals to 3.0.

In the approach by inelastic reserve capacity, the stress distribution in the section is governed by the compression strain factor  $(C_y)$ . If the factor is equal to 1.0, the stress distribution is linearly elastic up to the yield strength  $(f_y)$  at the extreme fibre of the compressive flange. If the factor is larger than 1.0, the location of the threshold of linear elastic stress distribution having the strain equal to yield strain  $(e_y)$ , beyond which a stress block of yield strength is formed in the section. Therefore, the stress distribution in the section is elastic-plastic. However, the calculation procedure of effective width is not clearly stated in the ASCE

Specification [7] and AS/NZS Standard [8] for sections having elastic-plastic stress distribution ( $C_y > 1$ ). Therefore, Yu and Laboube [12] suggested that the effective width equations for sections with linear elastic stress distribution can be used for those having elastic-plastic stress distribution. However, such design calculation leads to a quite conservative prediction for stocky members, especially those with  $C_y$  equals to 3.0.

In this study, the modified design rules for moment capacities  $(M^{\#}_{inelastic})$  consist of three parts, namely for (i) effective width calculation; (ii) upper bound limit of moment capacity; and (iii) limitation of web slenderness ( $d_w/t$  ratio). Firstly, the effective width calculation for specimens with compression strain factor  $(C_v)$  larger than 1.0, it is recommended that the height of the stress gradient in the compression portion of the web  $(d_g)$ , as shown in Fig. 7(b), is assumed to be fully effective, while that of the effective width for the plastic compressive stress block is calculated by the equations for webs with uniform compression, as indicated in the Clause 2.2.1 of the ASCE Specification [7] and AS/NZS Standard [8]. The effective width and the stress distribution in the sections calculated by Yu and Laboube [12] method and the proposed method are shown in Figs 7(a) and 7(b), respectively. Secondly, the upper bound limit of moment capacity from the approach by inelastic reserve capacity shall not exceed 1.25 times that obtained from the approach by initiation of yielding ( $M_{inelastic} \leq$  $1.25M_{vielding}$ ). Hence, this provides conservative predictions for specimens with low  $d/(t\varepsilon)$ value, due to the overly conservative predictions by initiation of yielding approach. Therefore it is recommended that the requirement of  $M_{inelastic} \le 1.25 M_{yielding}$  is not required in designing lean duplex stainless steel flexural members. Thirdly, the limitation of web slenderness of  $d_w/t < \lambda_l$  in calculating the moment capacity can be removed. It is shown in Table 5 that the approach by inelastic reserve capacity provided good predictions to the 47 specimens exceeded the web slenderness limit  $(d_w/t \ge \lambda_l)$ .

The comparison of the test and numerical results with the predications from the modified design rules using the inelastic reserve capacity approach is summarized in Table 5 and Fig. 6. The mean value of the  $M_u/M_{inelastic}^{\#}$  ratio equals to 1.18 with COV of 0.116, and is considered to be reliable with the reliability index ( $\beta_0$ ) of 2.98 and 2.79 for ASCE Specification and AS/NZS Standard, respectively. Fig. 6 shows the improvement of moment capacity predictions by the modified design rules, especially for those specimens with compression strain factor larger than 1.0. It is also observed from Table 6 that the accuracy of the prediction to the specimens with compression strain factor ( $C_y$ ) equals to 3.0 improves considerably compared to the current design rule. The mean value of the  $M_u/M_{inelastic}^{\#}$  ratio equals to 1.20 with the COV of 0.119, and the reliability index ( $\beta_0$ ) of 3.01 and 2.82 for

ASCE Specification and AS/NZS Standard, respectively. The modified design rules for moment capacities ( $M^{\#}_{inelastic}$ ) are validated against the numerical and experimental data of specimens with B/t ratio ranged from 6.25 to 140, and D/t ratio ranged from 6.25 to 83.3.

#### 8.3 EC3 Code

The rotational capacities (R) of the specimens, where the ultimate moments ( $M_u$ ) are larger than the plastic moment ( $M_{pl}$ ), are plotted against the web slenderness ( $d/t\varepsilon$ ) and flange slenderness ( $b/t\varepsilon$ ) in Figs 10 and 11. As discussed in Section 7.3 of this paper, the Class 1 limits for carbon steel in EC3 Part 1.1 [13] are more appropriate than those in EC3 Part 1.4 [9] for stainless steel. In addition, the moment capacities of the flange governed and web governed specimens that normalized with the elastic and plastic moment capacities are shown in Figs 12 – 15. It is shown that the Class 2 and Class 3 limits for carbon steel in EC3 Part 1.1 [13] are more appropriate for the cold-formed lean duplex stainless steel flexural members. Therefore, it is recommended that the Class 1, Class 2 and the Class 3 limits for carbon steel in EC3 Part 1.1 [13] are adopted for lean duplex stainless steel flexural members. Therefore, it is suggested that the class limits ( $b/t\varepsilon$ ) of 33, 38 and 42 are used for Classes 1, 2 and 3 for element subject to bending, respectively.

The effective width formula for internal element is modified and shown in Eq. (5):

$$\rho = \frac{0.7}{\overline{\lambda}_p} - \frac{0.04}{\overline{\lambda}_p^2} \le 1 \tag{5}$$

The design strengths calculated by the proposed class limits and effective width formula are represented by  $M_{EC3}^{\#}$ , and the comparison of the test and numerical results with the design strengths predicted by the modified design rules is summarized in Table 5 and Fig. 8. The mean value of the  $M_u/M_{EC3}^{\#}$  ratio equals to 1.15 and the COV equals to 0.123. It is shown that the modified design rules are less conservative than the current EC3 Code predictions. The EC3 modified design rules are considered to be reliable with the reliability index ( $\beta_0$ ) equal to 2.65. Table 6 shows that the experimental and numerical results-to-prediction moment ratio of Classes 1, 2 and 3 sections reduced from 1.35 to 1.23, while that of the Class 4 sections reduced from 1.17 to 1.09 when the current design predictions compared with the modified design predictions. The EC3 modified design rules are considered to be reliable for all four Classes 1, 2, 3 and 4 sections, with the reliability index larger than or equal to the target value

of 2.50. The modified design rules are validated against the numerical and experimental data of specimens with D/t ratio ranged from 6.25 to 140, and B/t ratio ranged from 6.25 to 83.3.

#### 8.4 Direct Strength Method

The nominal flexural strength ( $M_{DSM}$ ) in the direct strength method [10] as calculated by Eq. (4) shows that the nominal flexural strength is equal to a constant value of yield moment ( $M_{DSM} = M_y$ ), when  $\lambda_l$  is less than or equal to 0.776 ( $\lambda_l \le 0.776$ ). In this study, based on a total number of 81 data with  $\lambda_l$  less than or equal to 0.776, it is shown that the flexural strengths generally decrease linearly as  $\lambda_l$  increases. Furthermore, the flexural strength predictions using the current direct strength method are generally conservative. Therefore, it is recommended to modify the current direct strength equation [10] to Eq. (6) as shown below:

$$M_{DSM}^{\#} = \begin{cases} 1.1((0.776 - \lambda_l) + 1)M_y & \text{for } \lambda_l \le 0.776 \\ \left(1 - 0.15\left(\frac{M_{crl}}{1.1M_y}\right)^{0.4}\right)\left(\frac{M_{crl}}{1.1M_y}\right)^{0.4} & \text{for } \lambda_l \le 0.776 \end{cases}$$
(6)

The comparison of the experimental and numerical data with the design values calculated by the modified DSM in Eq. (6) is shown in Tables 5-6 and Fig. 16. The design strengths calculated by the modified DSM are represented by  $M_{DSM}^{\#}$ . The mean value of the  $M_u/M_{DSM}^{\#}$ is 1.07 with COV of 0.127 and the reliability index ( $\beta_0$ ) of 2.55, as shown in Table 5. It is shown that the modified DSM provides more accurate and less scattered predictions compared to the current DSM for the cold-formed lean duplex stainless steel flexural members. The comparison of the experimental and numerical data with the design values for the 81 specimens with  $\lambda_l$  less than or equal to 0.776, and the 99 specimens with  $\lambda_l$  greater than 0.776, is summarized in Table 6. The mean values of  $M_u/M^{\#}_{DSM}$  for specimens with  $\lambda_l \leq$ 0.776 and  $\lambda_l > 0.776$  are equal to 1.04 and 1.07 with COV of 0.062 and 0.142, respectively. The modified DSM is considered to be reliable for these two groups of specimens with reliability index ( $\beta_0$ ) of 2.70 and 2.50, respectively. It is shown that the modified DSM provides the most accurate predictions among the design rules discussed earlier. Furthermore, the tedious iterative process is not required. The modified direct strength equation in Eq. (6) is validated against the experimental and numerical data of specimens with  $\lambda_l$  ranged from 0.13 to 2.0.

### 9. Conclusions

Experimental and numerical investigation on the structural performance of cold-formed lean duplex stainless steel flexural members has been presented in this paper. A series of four-point bending tests was conducted on square and rectangular hollow sections. A finite element model of flexural members was developed and verified with the experimental results. A wide range of parametric study was performed using the verified finite element model. The experimental and numerical results obtained from this study and the available data were compared with the design strengths predicted by the American Specification [7], Australian/New Zealand Standard [8], European Code [9], modified European Code by Gardner and Theofanous [2], direct strength method [10] and continuous strength method [6]. It is shown that the continuous strength method is capable of providing good prediction for cold-formed lean duplex stainless steel square and rectangular hollow sections flexural members, while the other design rules are quite conservative. Modifications on the design rules in the AS/NZS, ASCE, EC3 and DSM are proposed and compared with the experimental and numerical results. It is shown that the modified design rules provide more accurate and less scatter predictions than the existing design rules. The modified direct strength method provides the most accurate predictions compared to the other design rules, and this method is relatively simple in calculating the flexural strengths. Therefore, it is recommended that the modified DSM is used in designing cold-formed lean duplex stainless steel flexural members of square and rectangular hollow sections. The modified DSM design equations are capable of producing reliable limit state designs when calibrated with resistance factor of 0.9.

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# Notation

The following symbols are used in this paper:

A	Full area
В	Overall width of the flange
b	Flat width of the flange
$b_e$	Effective width of compressive flange
$C_{y}$	Compression strain factor in American Specification and Australian/New Zealand
	Standard
D	Overall depth of the web
d	Flat portion of the web
$d_e$	Effective width of stress block in the compressive web
$d_{e1}$	Portion of the effective width of compressive web in stress gradient
$d_{e2}$	Portion of the effective width of compressive web in stress gradient
$d_g$	Height of the stress gradient in the compression portion of the web in the
	modified approach by inelastic reserve capacity
$d_w$	Depth of the compressed portion of the web
$E_o$	Initial Young's modulus
$e_y$	Yield strain in American Specification and Australian/New Zealand Standard
$F_m$	Mean value of fabrication factor
$f_{y}$	Yield strength
k	Curvature
$k_{Exp,u}$	Curvature corresponding to the experimental ultimate moment
$k_{FEA,u}$	Curvature corresponding to the ultimate moment predicted by finite element
	analysis
$k_{pl}$	Curvature corresponding to the plastic moment $(M_{pl})$ on the ascending branch of
	moment-curvature curve
$k_u$	Curvature at ultimate moment
$k^{}_{pl}$	Curvature corresponding to the plastic moment $(M_{pl})$ on the descending branch of
	moment-curvature curve
L	Length of specimen
$M_{crl}$	Critical elastic local buckling moment
$M_{DSM}$	Unfactored design moment capacity predicted by the direct strength method
$M^{\#}_{DSM}$	Unfactored design moment capacity predicted by the modified direct strength method

 $M_d$ Moment capacities predicted by design rules  $M_{EC3}$ Unfactored design moment capacity predicted by the European Code  $M_{Exp}$ Experimental ultimate moment (test moment capacity) Ultimate moment predicted by finite element analysis  $M_{FFA}$  $M_{el}$ Elastic bending moment  $M^{\#}_{EC3}$ Unfactored design moment capacity predicted by the modified European Code Unfactored design moment capacity predicted by the modified European Code by  $M_{G\&T}$ Gardner and Theofanous Unfactored design moment capacity predicted by the approach by inelastic  $M_{inelastic}$ reserve capacity in American Specification and Australian/New Zealand Standard Unfactored design moment capacity predicted by the approach by inelastic M<sup>\*</sup>inelastic reserve capacity for specimens with the ratio of the depth of the compressed portion of the web to its thickness exceeded the slenderness ratio  $M^{\#}_{inelastic}$ Unfactored design moment capacity predicted by the modified approach by inelastic reserve capacity  $M_m$ Mean value of material factor  $M_{ne}$ Nominal flexural strength for lateral-torsional buckling in direct strength method Nominal flexural strength for local buckling in direct strength method  $M_{nl}$ Nominal flexural strength for distortional buckling in direct strength method  $M_{nd}$  $M_{pl}$ Plastic bending moment Experimental and numerical ultimate moments  $M_{u}$  $M_{\rm v}$ Yield moment Unfactored design moment capacity predicted by the approach by initiation of  $M_{yielding}$ yielding in American Specification and Australian/New Zealand Standard Ramberg-Osgood parameter n Mean value of tested-to-predicted load ratio  $P_m$ R Rotational capacity Radius of the curved beam specimen between the LVDTs located at the two r loading points Inner radius  $r_i$ Outer radius  $r_o$ Effective section modulus  $S_e$  $S_f$ Gross section modulus t Thickness of specimen

Coefficient of variation of fabrication factor

Coefficient of variation of material factor

 $V_F$ 

 $V_m$ 

$V_p$	Coefficient of variation of tested-to-predicted load ratio
$oldsymbol{eta}_0$	Reliability index
$oldsymbol{eta_{I}}$	Reliability index
$\mathcal{E}$	Material factor in European Code
$\mathcal{E}_{\!f}$	Tensile strain after fracture based on gauge length of 25 mm
$\phi_0$	Resistance factor
$\phi_{I}$	Resistance factor
$\lambda_l$	Slenderness ratio in American Specification, Australian/New Zealand Standard
	and direct strength method
$\overline{\lambda}_{p}$	Element slenderness in European Code
$\rho$	Value in calculating effective area in American Specification, Australian/New
	Zealand Standard and European Code
$\sigma_{0.2}$	0.2% tensile proof stress
$\sigma_{\!\scriptscriptstyle u}$	Tensile strength

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Section		F	Flat				Co	orner		
$(D\times B\times t)$	$\sigma_{0.2}$	$\sigma_u$	$\mathcal{E}_f$	$E_o$	n	$\sigma_{0.2}$	$\sigma_u$	$\mathcal{E}_f$	$E_o$	n
	(MPa)	(MPa)	(%)	(GPa)		(MPa)	(MPa)	(%)	(GPa)	
$50 \times 30 \times 2.5$	683	764	39	198	7	788	975	22	192	4
$50 \times 50 \times 1.5$	610	734	44	194	5	824	1012	15	200	5
$50 \times 50 \times 2.5$	635	756	44	198	6	833	1079	19	207	5
$70 \times 50 \times 2.5$	613	738	44	194	8	844	995	21	200	5
$100 \times 50 \times 2.5$	625	727	49	200	6	882	1033	17	203	5
150×50×2.5	664	788	35	202	4	831	967	18	199	6

Table 1: Measured material properties obtained from tensile coupon tests [3]

Specimen	D	В	t	$r_o$	$r_i$	L	A
$(D\times B\times t)$	(mm)	(mm)	(mm)	(mm)	(mm)	(mm)	$(mm^2)$
50×30×2.5L900	50.0	30.0	2.571	5.0	2.0	900	364.2
30×50×2.5L900	30.3	50.5	2.574	5.0	2.0	900	359.7
50×50×1.5L900	50.2	50.2	1.527	3.0	1.0	900	290.2
50×50×2.5L900	50.0	50.1	2.487	3.0	1.0	900	457.4
70×50×2.5L1100	70.5	50.8	2.553	4.0	1.0	1100	575.1
50×70×2.5L1100	50.7	70.4	2.513	4.0	1.0	1100	565.7
100×50×2.5L1500	100.4	50.4	2.497	3.5	1.0	1500	713.9
50×100×2.5L1500	50.7	100.1	2.480	3.5	1.0	1500	708.3
150×50×2.5L1500	150.4	50.1	2.473	4.5	2.0	1500	950.0
50×150×2.5L1500	50.1	149.9	2.459	4.5	2.0	1500	944.7

Table 2: Measured specimen dimensions

Specimen $(D \times B \times t)$	M <sub>Exp</sub> (kNm)	$k_{Exp,u} \times 10^{-4}$ (mm <sup>-1</sup> )	Failure mode	M <sub>FEA</sub> (kNm)	$k_{FEA,u} \times 10^{-4}$ (mm <sup>-1</sup> )	Failure mode	$\frac{M_{Exp}}{M_{FEA}}$	$\frac{M_{Exp}}{M_{el}}$	$\frac{M_{Exp}}{M_{pl}}$	$\frac{k_{Exp,u}}{k_{FEA,u}}$
50×30×2.5L900	5.4	19.6	F	5.2	20.1	F	1.04	1.66	1.29	0.97
30×50×2.5L900	3.4	10.8	F	3.5	11.6	F	0.97	1.40	1.15	0.93
50×50×1.5L900	3.3	2.7	L+F	3.6	2.3	L+F	0.93	1.19	1.01	1.17
50×50×2.5L900	7.0	5.2	F	6.9	6.0	F	1.02	1.60	1.34	0.87
$70 \times 50 \times 2.5 \text{L}1100$	10.8	3.7	F	10.4	4.7	F	1.04	1.50	1.23	0.80
50×70×2.5L1100	7.7	3.7	L+F	7.6	3.1	L+F	1.01	1.31	1.12	1.16
100×50×2.5L1500	17.8	2.8	F	17.1	2.7	F	1.04	1.52	1.23	1.04
50×100×2.5L1500	8.3	2.7	L+F	8.4	1.8	L+F	0.99	1.05	0.93	1.50
150×50×2.5L1500	29.9	1.1	F	31.6	1.3	F	0.94	1.33	1.03	0.85
50×150×2.5L1500	9.1	2.2	L+F	9.1	2.1	L+F	1.00	0.76	0.68	1.08
	•			•		Mean	1.00	1.33	1.10	1.04
						COV	0.040	0.206	0.178	0.199

Table 3: Comparison of test strengths with finite element analysis results

Specimen	$M_{FEA}$	$k_{FEA,u} \times 10^{-4}$	Specimen	$M_{FEA}$	$k_{FEA,u}\times10^{-4}$
$(D\times B\times t)$	(kNm)	(mm <sup>-1</sup> )	$(D\times B\times t)$	(kNm)	(mm <sup>-1</sup> )
50×50×6L900	14.2	21.8	100×50×8L900	53.6	10.2
$50 \times 50 \times 4L900$	9.7	18.4	100×50×5L900	33.9	14.5
$50 \times 50 \times 2L900$	4.7	4.6	100×50×3L900	19.5	3.7
$50 \times 50 \times 1.5 L900$	3.2	1.8	100×50×2L900	11.8	1.3
50×50×1L900	1.8	1.6	100×50×1L900	3.7	0.7
100×100×10L900	95.5	7.2	50×100×8L900	28.2	8.7
100×100×5L900	47.7	2.6	50×100×5L900	18.4	5.4
100×100×3L900	24.7	1.1	50×100×3L900	10.2	1.9
$100 \times 100 \times 2.5 L900$	19.3	1.0	50×100×2.5L900	7.8	1.8
100×100×1.5L900	8.6	0.6	50×100×2L900	6.0	1.5
150×150×15L1500	323.0	5.1	50×100×1.5L900	3.7	1.3
150×150×10L1500	217.8	3.4	300×100×10L2100	503.3	1.9
150×150×5L1500	96.8	0.9	300×100×8L2100	386.1	1.1
150×150×4L1500	71.9	0.7	300×100×7.5L2100	354.8	1.0
150×150×3L1500	44.4	0.5	300×100×7L2100	324.1	0.9
200×200×20L2100	766.8	4.1	300×100×6.5L2100	306.3	0.8
200×200×10L2100	384.5	1.8	300×100×6L2100	276.7	0.7
200×200×6L2100	198.4	0.6	300×100×5.5L2100	244.4	0.5
200×200×4L2100	101.5	0.4	300×100×5L2100	218.5	0.5
200×200×2.5L2100	49.0	0.3	300×100×4L2100	144.0	0.3
250×250×20L2100	1207.0	2.5	300×100×3L2100	73.4	0.2
250×250×15L2100	908.0	1.9	100×300×15L1500	285.5	2.3
250×250×8L2100	424.4	0.6	100×300×10L1500	149.6	1.1
250×250×6L2100	274.2	0.4	100×300×8L1500	133.7	0.7
250×250×4L2100	146.1	0.3	100×300×6L1500	90.2	0.7
300×300×25L3000	2168.0	2.3	100×300×5.5L1500	80.3	0.5
300×300×15L3000	1290.0	1.2	100×300×5L1500	70.8	0.5
300×300×105L3000	780.0	0.5	100×300×4L1500	48.4	0.5
300×300×7L3000	460.6	0.4	350×100×15L3000	988.0	4.2
300×300×5L3000	275.0	0.3	350×100×10L3000	637.0	1.4
70×50×8L900	30.0	26.4	350×100×8L3000	493.9	0.8
$70 \times 50 \times 4L900$	15.9	14.9	350×100×7.5L3000	459.2	0.7
$70 \times 50 \times 2L900$	7.3	2.6	350×100×7L3000	414.5	0.5
70×50×1.5L900	5.0	1.4	350×100×6.5L3000	382.8	0.5
70×50×1L900	2.5	1.2	350×100×6L3000	344.1	0.5
50×70×8L900	17.3	9.2	350×100×5.5L3000	302.6	0.4
50×70×4L900	11.7	6.4	350×100×5L3000	264.5	0.3
50×70×2.5L900	6.8	3.1	350×100×4L3000	183.2	0.3
50×70×2L900	5.0	1.9	100×350×15L1500	312.9	1.1
50×70×1.5L900	3.4	1.9	100×350×12L1500	242.2	1.0
50×70×1L900	1.8	1.3	100×350×9L1500	163.7	0.6
50×30×4L900	7.7	37.1	100×350×7L1500	116.5	0.7
	5.7	34.5	100×350×6.5L1500	104.9	0.6
うひ×うひ×うL9ひひ					
50×30×3L900 50×30×2L900	3.7	19.6	100×350×6L1500	94.2	0.5

50×30×1L900         1.6         2.2         280×70×8L2100         313.0         1.5           30×50×4L900         4.5         28.2         280×70×6L2100         221.8         0.8           30×50×3L900         3.5         11.9         280×70×5.5L2100         200.8         0.8           30×50×2L900         2.4         7.8         280×70×5L2100         176.3         0.6           30×50×1.5L900         1.6         3.1         280×70×4.5L2100         152.9         0.5           30×50×1L900         0.9         3.6         280×70×4L2100         127.0         0.4           250×100×15L2100         578.1         7.7         280×70×3.5L2100         103.4         0.3           250×100×8L2100         301.7         2.4         280×70×3L2100         78.9         0.3           250×100×6L2100         217.3         1.2         70×280×11L900         123.2         1.9           250×100×5.5L2100         195.6         1.0         70×280×9L900         98.6         1.4           250×100×5L2100         174.0         0.8         70×280×8L900         84.9         1.3           250×100×4L2100         126.5         0.5         70×280×5L900         41.0         0.8           100×250						
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	50×30×1L900	1.6	2.2	280×70×8L2100	313.0	1.5
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	30×50×4L900	4.5	28.2	280×70×6L2100	221.8	0.8
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$	30×50×3L900	3.5	11.9	280×70×5.5L2100	200.8	0.8
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$	$30 \times 50 \times 2L900$	2.4	7.8	280×70×5L2100	176.3	0.6
250×100×15L2100         578.1         7.7         280×70×3.5L2100         103.4         0.3           250×100×10L2100         383.5         5.2         280×70×3L2100         78.9         0.3           250×100×8L2100         301.7         2.4         280×70×2L2100         47.6         0.1           250×100×6L2100         217.3         1.2         70×280×11L900         123.2         1.9           250×100×5.5L2100         195.6         1.0         70×280×9L900         98.6         1.4           250×100×5L2100         174.0         0.8         70×280×8L900         84.9         1.3           250×100×4L2100         126.5         0.5         70×280×6L900         58.6         0.8           100×250×13L1500         221.2         2.6         70×280×5L900         41.0         0.8	30×50×1.5L900	1.6	3.1	280×70×4.5L2100	152.9	0.5
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	30×50×1L900	0.9	3.6	280×70×4L2100	127.0	0.4
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	250×100×15L2100	578.1	7.7	280×70×3.5L2100	103.4	0.3
250×100×6L2100       217.3       1.2       70×280×11L900       123.2       1.9         250×100×5.5L2100       195.6       1.0       70×280×9L900       98.6       1.4         250×100×5L2100       174.0       0.8       70×280×8L900       84.9       1.3         250×100×4L2100       126.5       0.5       70×280×6L900       58.6       0.8         100×250×13L1500       221.2       2.6       70×280×5L900       41.0       0.8	250×100×10L2100	383.5	5.2	280×70×3L2100	78.9	0.3
250×100×5.5L2100       195.6       1.0       70×280×9L900       98.6       1.4         250×100×5L2100       174.0       0.8       70×280×8L900       84.9       1.3         250×100×4L2100       126.5       0.5       70×280×6L900       58.6       0.8         100×250×13L1500       221.2       2.6       70×280×5L900       41.0       0.8	250×100×8L2100	301.7	2.4	280×70×2L2100	47.6	0.1
250×100×5L2100       174.0       0.8       70×280×8L900       84.9       1.3         250×100×4L2100       126.5       0.5       70×280×6L900       58.6       0.8         100×250×13L1500       221.2       2.6       70×280×5L900       41.0       0.8	250×100×6L2100	217.3	1.2	70×280×11L900	123.2	1.9
250×100×4L2100       126.5       0.5       70×280×6L900       58.6       0.8         100×250×13L1500       221.2       2.6       70×280×5L900       41.0       0.8	250×100×5.5L2100	195.6	1.0	70×280×9L900	98.6	1.4
100×250×13L1500 221.2 2.6 70×280×5L900 41.0 0.8	250×100×5L2100	174.0	0.8	70×280×8L900	84.9	1.3
	250×100×4L2100	126.5	0.5	70×280×6L900	58.6	0.8
$100\sqrt{250}\sqrt{01}\ 1500$ $149.7$ $1.2$ $70\sqrt{290}\sqrt{4}\ 51.000$ $26.5$ $0.6$	100×250×13L1500	221.2	2.6	70×280×5L900	41.0	0.8
100×250×9L1500 146.7 1.2   70×260×4.5L900 50.5 0.0	100×250×9L1500	148.7	1.2	70×280×4.5L900	36.5	0.6
100×250×7L1500 110.6 0.9 70×280×4L900 28.0 0.5	100×250×7L1500	110.6	0.9	70×280×4L900	28.0	0.5
100×250×5L1500 66.0 0.6	100×250×5L1500	66.0	0.6			
100×250×4.5L1500 56.8 0.9	100×250×4.5L1500	56.8	0.9			
100×250×4L1500 47.7 0.6	100×250×4L1500	47.7	0.6			
100×250×3L1500 29.0 0.6	100×250×3L1500	29.0	0.6			

Table 4. Parametric study results

	ASCE					AS/NZS				EC3			DSM	
	$M_u$	$M_{u}$	$M_{u}$	$M_{u}$	$M_{u}$	$M_u$	$M_{u}$	$M_{u}$	$M_{u}$	$M_{u}$	$M_{u}$	$M_{u}$	$M_u$	$M_{u}$
	$M_{yielding}$	$\overline{M}_{inelastic}$	$\overline{M}^*_{inelastic}$	$M_{inelastic}^{\#}$	M yielding	$\overline{M}_{inelastic}$	$\overline{M}^*_{inelastic}$	$M_{inelastic}^{\#}$	$\overline{M}_{EC3}$	$\overline{M_{G\&T}}$	$\overline{M_{EC3}^{\#}}$	$\overline{M}_{DSM}$	$\overline{M_{DSM}^{\#}}$	$\overline{M}_{CSM}$
# of data	180	133	47*	180	180	133	47*	180	180	180	180	180	180	96
Mean $(P_m)$	1.50	1.34	1.17	1.18	1.50	1.34	1.17	1.18	1.25	1.19	1.15	1.35	1.07	1.13
$COV(V_p)$	0.264	0.200	0.153	0.116	0.264	0.200	0.153	0.116	0.147	0.133	0.123	0.172	0.127	0.087
Resistance factor ( $\phi_0$ )	0.90	0.90	0.90	0.90	0.90	0.90	0.90	0.90	0.91	0.91	0.91	0.90	0.90	0.91
Reliability index $(\beta_0)$	2.85	2.92	2.72	2.98	2.71	2.76	2.55	2.79	2.83	2.73	2.65	3.12	2.55	2.75
Resistance factor $(\phi_l)$	0.90	0.90	0.90	0.90	0.90	0.90	0.90	0.90	0.90	0.90	0.90	0.90	0.90	0.90
Reliability index $(\beta_I)$	2.85	2.92	2.72	2.98	2.85	2.92	2.72	2.98	3.00	2.91	2.83	3.12	2.55	2.94

Specimens that exceeded the limitation  $(d_w/t > \lambda_l)$  of the approach by inelastic reserve capacity in ASCE and AS/NZS

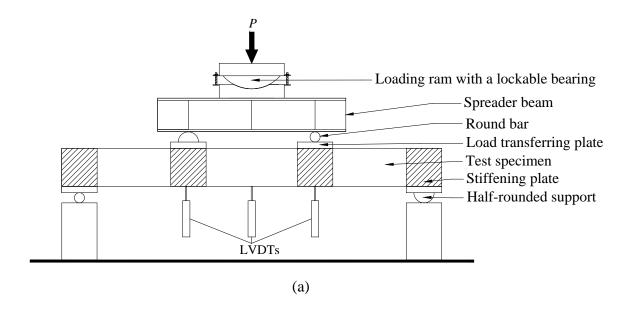
Table 5: Comparison of the tests and numerical results with the design moment capacities

<sup>\*</sup> Modified design rules

	AS	CE	AS/	NZS			Е	C3				DS	SM	
	$C_{y}$	= 3	$C_{y}$	= 3	(	Class 1, 2,	3		Class 4		$\lambda_l < 0$	0.776	$\lambda_l \geq$	0.776
	$M_{u}$	$M_{u}$	$M_{u}$	$M_{u}$	$M_{u}$	$M_{u}$	$M_{u}$	$M_{u}$	$M_{u}$	$M_{u}$	$M_{u}$	$M_{u}$	$M_{u}$	$M_{u}$
	$\overline{M}_{inelastic}$	$M_{inelastic}^{\#}$	$\overline{M}_{inelastic}$	$M_{inelastic}^{\#}$	$\overline{M}_{EC3}$	$\overline{M_{G\&T}}$	$\overline{M_{EC3}^{\#}}$	$\overline{M}_{EC3}$	$\overline{M_{G\&T}}$	$\overline{M_{EC3}^{\#}}$	$\overline{M}_{DSM}$	$M_{DSM}^{\#}$	$\overline{M}_{DSM}$	$\overline{M_{DSM}^{\#}}$
# of data	54	82	54	82	63	81	94	117	101	86	81	81	99	99
Mean $(P_m)$	1.54	1.20	1.54	1.20	1.35	1.47	1.23	1.17	1.11	1.09	1.47	1.04	1.16	1.07
$COV(V_p)$	0.179	0.119	0.179	0.119	0.083	0.138	0.085	0.154	0.135	0.110	0.138	0.062	0.084	0.142
Resistance factor $(\phi_0)$	0.90	0.90	0.90	0.90	0.91	0.90	0.91	0.91	0.91	0.91	0.90	0.90	0.90	0.90
Reliability index $(\beta_0)$	3.50	3.01	3.33	2.82	3.45	3.50	3.07	2.57	2.46	2.50	3.50	2.70	3.04	2.50
Resistance factor $(\phi_I)$	0.90	0.90	0.90	0.90	0.90	0.90	0.90	0.90	0.90	0.90	0.90	0.90	0.90	0.90
Reliability index $(\beta_l)$	3.50	3.01	3.50	3.01	3.64	3.64	3.26	2.75	2.64	2.69	3.64	2.70	3.04	2.50

<sup>\*</sup> Modified design rules

Table 6: Comparison of the tests and numerical results with the design moment capacities



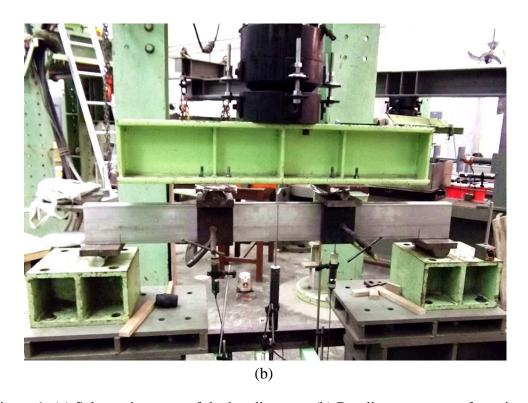
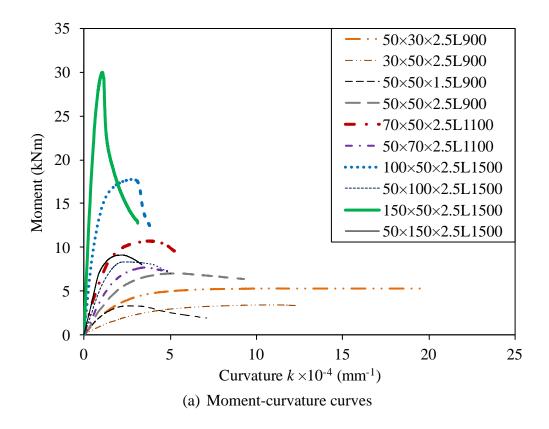


Figure 1: (a) Schematic set up of the bending test. (b) Bending test setup of specimen  $150\times50\times2.5L1500$ 



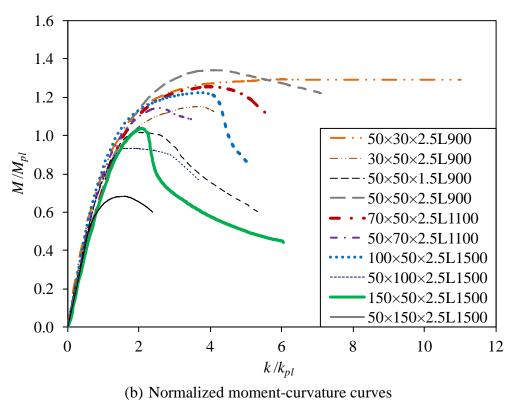


Figure 2: Moment-curvature curves of test specimens



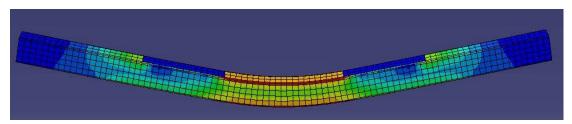


Figure 3. Experimental and numerical failure mode of flexural failure for specimen 50×30×2.5L900



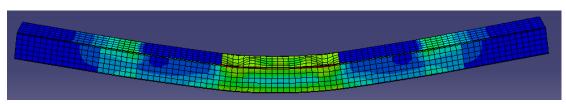


Figure 4. Experimental and numerical failure mode of interaction of local and flexural failure for specimen 50×50×1.5L900

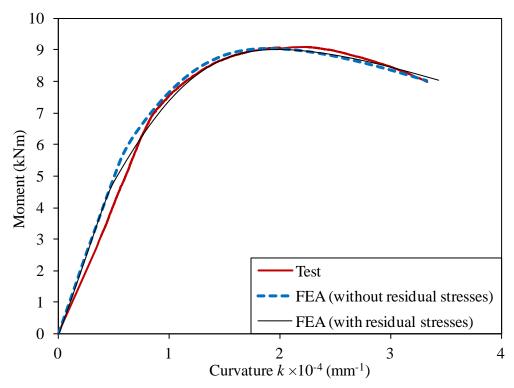


Figure 5: Comparison of load-displacement curves of test and FEA for specimen  $50 \times 150 \times 2.5 L1500$ 

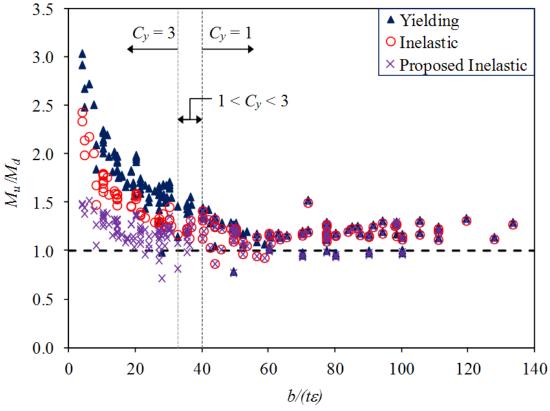
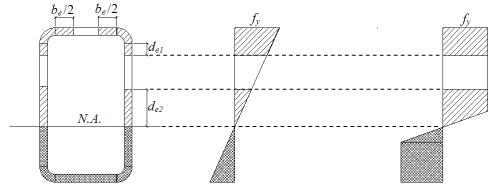


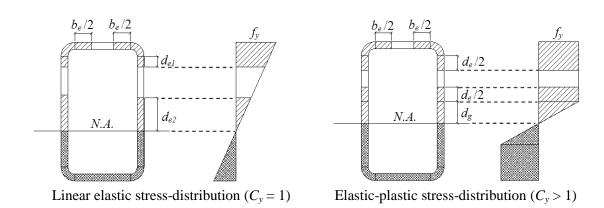
Figure 6: Comparison of test and numerical results with design strengths by ASCE and AS/NZS



Linear elastic stress-distribution ( $C_y = 1$ )

Elastic-plastic stress-distribution  $(C_y > 1)$ 

# (a) Yu & Laboube [12]



# (a) Proposed

Figure 7: Effective width of cross-section with linear elastic and elastic plastic stress distribution

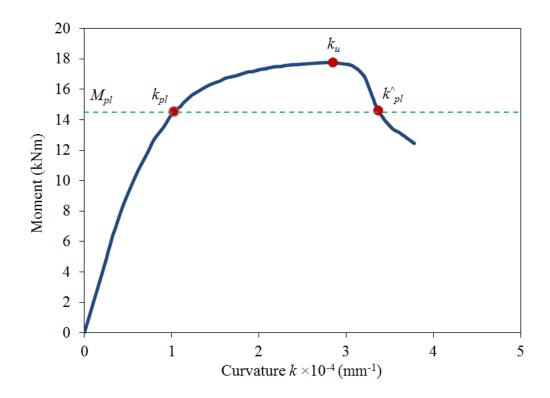


Figure 8: Locations of  $k_{pl}$ ,  $k_{pl}$  and  $k_u$  on moment-curvature curve of specimen  $100 \times 50 \times 2.5 \text{L} 1500$ 

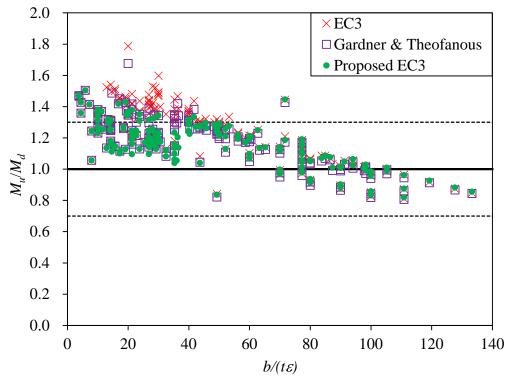


Figure 9: Comparison of tests and numerical results with design strengths by EC3

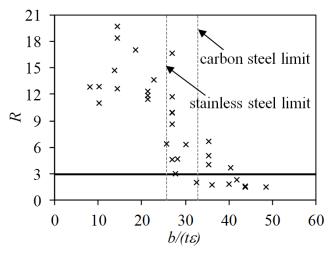


Figure 10: Class 1 limit of element subject to compression

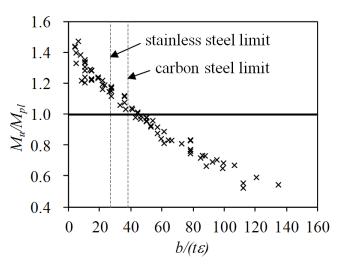


Figure 12: Class 2 limit of element subject to compression

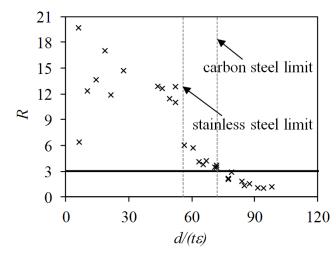


Figure 11: Class 1 limit of element subject to bending

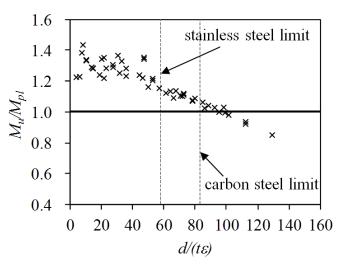


Figure 13: Class 2 limit of element subject to bending

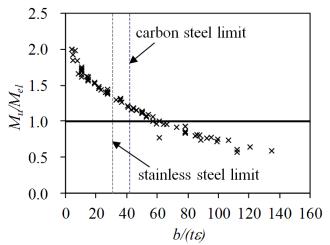


Figure 14. Class 3 limit of element subject to compression

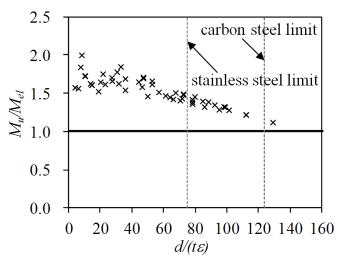


Figure 15. Class 3 limit of element subject to bending

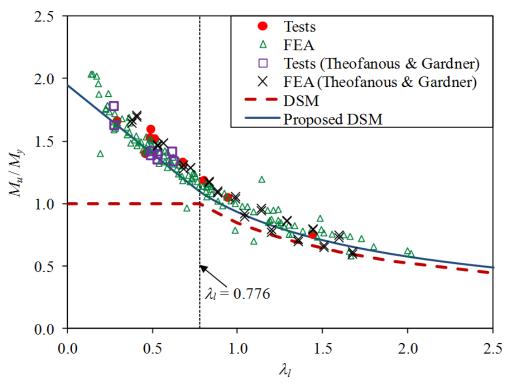


Figure 16: Comparison of tests and numerical results with design strengths by DSM  $M_u/M_y$